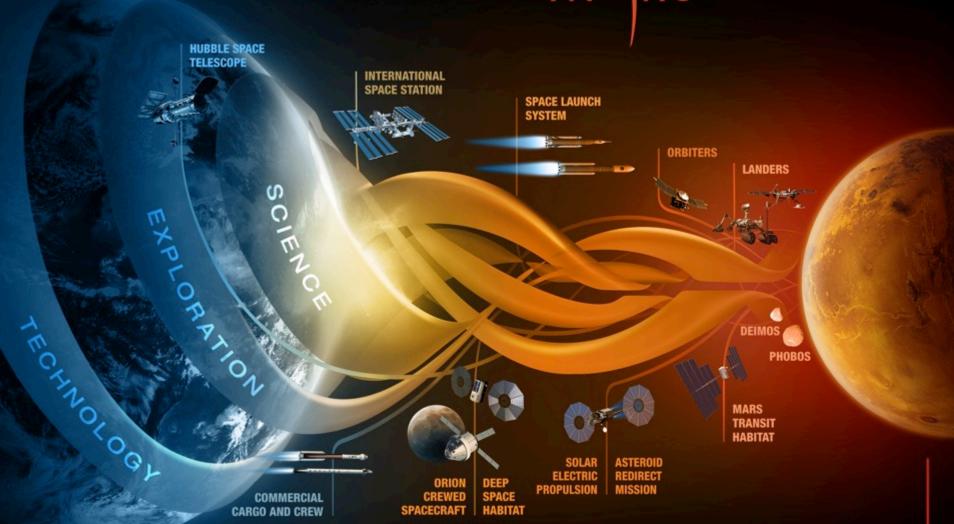


# JOURNEY TO MARS





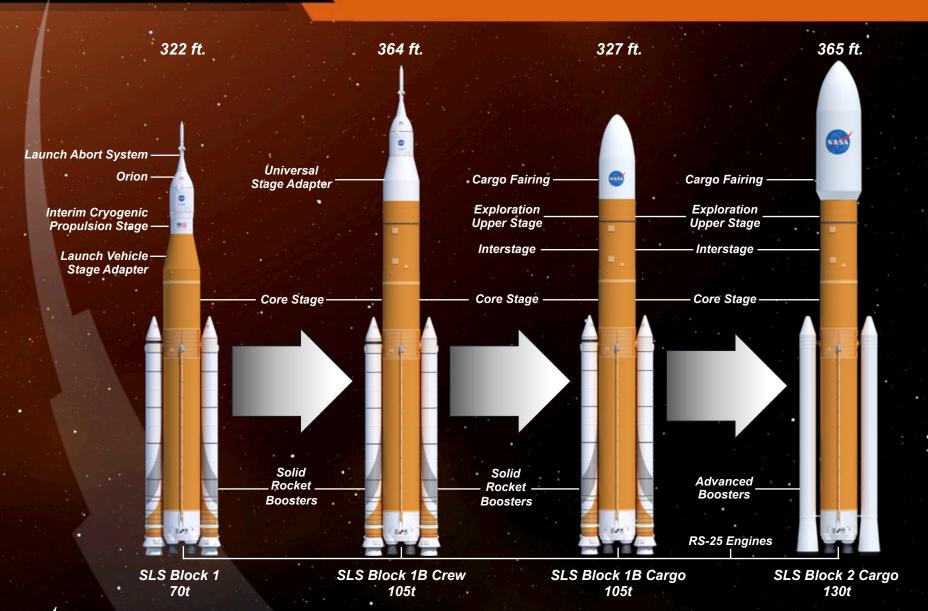
MISSIONS: 6-12 MONTHS RETURN: HOURS

EARTH RELIANT

MISSIONS: 1-12 MONTHS
RETURN: DAYS
PROVING GROUND

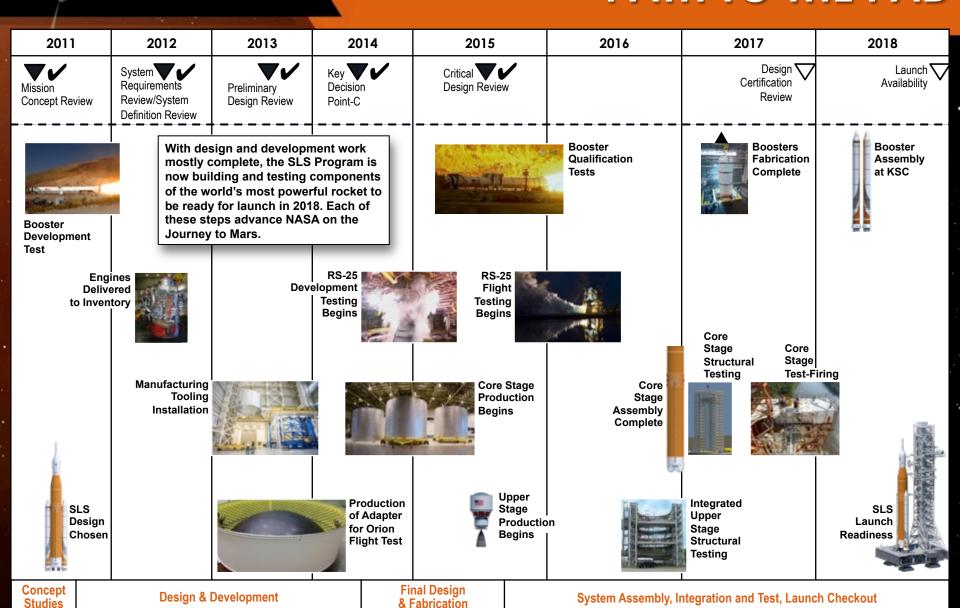
MISSIONS: 2–3 YEARS RETURN: MONTHS EARTH INDEPENDENT

# **SLS EVOLUTION OVERVIEW**





### PATH TO THE PAD



SLS SPACE LAUNCH SYSTEM

## **CORE STAGE**

### **OVERVIEW**

- World's largest rocket stage
- 27.6-foot diameter; 200 feet
- tall
- Being built at Michoud Assembly Facility outside New Orleans, LA



### **STATUS**

- Welding is underway currently on test and flight articles for core stage fuel tanks
- Refurbishment underway on B-2 stand at Stennis for Green Run core stage test





# **CORE STAGE**



# **RS-25 CORE STAGE ENGINES**

### **OVERVIEW**

- World's most powerful, efficient and reliable liquid rocket engine
- Four Space Shuttle-heritage RS-25s
- Upgraded with new controller; engines certified at 512,000 pounds of thrust each

### **STATUS**

- Sixteen flight engines currently in inventory
- SLS RS-25 testing began at Stennis Space Center in early 2015; currently ongoing



# LIQUID MAIN ENGINES



## SOLID ROCKET BOOSTERS

### **OVERVIEW**

- World's most powerful solid boosters for flight
- Two Space Shuttle-heritage solid rocket boosters
- Upgraded via fifth propellant segment to 3.6 million pounds of thrust capability, and with new avionics and insulation

### **STATUS**

- Second Qualification Motor test completed in June 2016
- Flight hardware in inventory at Kennedy Space Center; processing underway at Orbital ATK in Utah





# FIVE-SEGMENT SOLID ROCKET BOOSTER



### **UPPER STAGE AND ADAPTERS**

### **OVERVIEW**

- Interim Cryogenic Propulsion Stage is derived from proven second stage of Delta IV Heavy
- Launch Vehicle Stage Adapter and Orion Stage Adapter mate ICPS to core stage and Orion, respectively

### **STATUS**



- Orion Stage Adapter became first original SLS hardware to fly on Exploration Flight Test-1 in December 2014
- Flight in manufacture currently; test articles will begin stacking for loads testing in late 2016



# **UPPER STAGE AND ADAPTERS**



# UPPER STAGE AND ADAPTERS



# SECONDARY PAYLOAD CAPABILITY

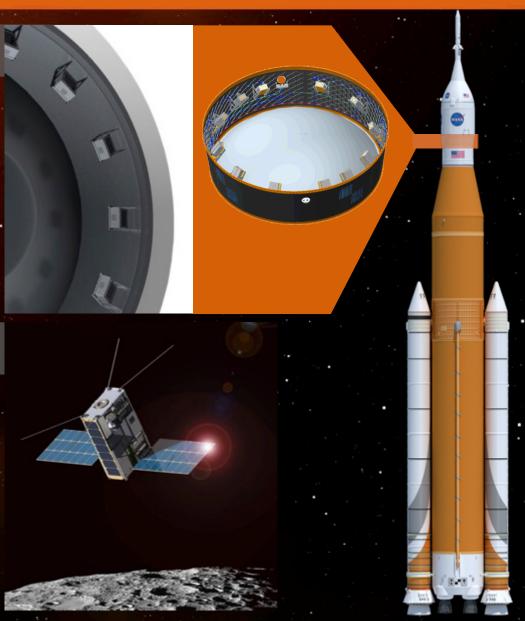
### **Accommodations**

SLS for Exploration Mission-1 will include thirteen 6U payload locations

6U volume/mass is the current standard (14 kg payload mass)

### **Future Capability**

 Options for larger secondary payloads being evaluated

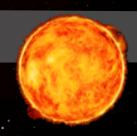


# ONE LAUNCH, MULTIPLE DISCIPLINES

Along with Orion, the first SLS launch will carry 13 6U smallsats, representing multiple disciplines and partners. The smallsats will be deployed from the Orion Stage Adapter.

### SUN

 CuSP (Southwest Research Institute)



### MOON

- Lunar Flashlight (NASA)
- Lunar IceCube (Morehead State University)
- LunaH-Map (Arizona State University)
- Omotenashi (JAXA)

#### **EARTH**

- EQUULEUS (JAXA)
- Skyfire (Lockheed Martin)

### AND BEYOND

- Biosentinel (NASA)
- ArgoMoon (ESA/ASI)
- Three Centennial Challenge Winners (TBD)



### **ASTEROID**

NEA Scout

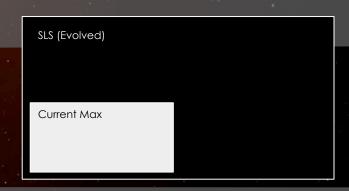




# BENEFITS OF SPACE LAUNCH SYSTEM

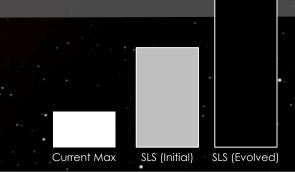
### **VOLUME**

- Space Launch System will be able to offer payload accommodations with five times more volume than any contemporary launch vehicle
- Payload fairings of up to 10-meter diameter are planned



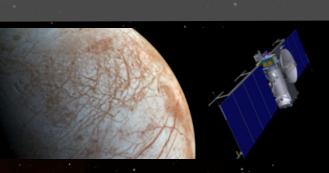
### MASS

- Space Launch System will offer an initial capability of greater than 70 metric tons to low Earth orbit; current U.S. launch vehicle maximum is 28 t
- Evolved version of SLS will offer Mars-enabling capability of greater than 130 t to LEO



### **DEPARTURE ENERGY**

- SLS offers reduced transit times to the outer solar system by half or greater
- Higher characteristic energy (C3) also enables larger payloads to destination





# THE ADVENTURE BEGINS NOW.

